

ENGINEERING

BASIC THEORY

Thermodynamics (Basic heat technology)

Isothermal condition/ phase	$Pv = \text{constant}$	P pressure v volume	Pa m^3
Adiabatic condition/ phase	$Pv^\kappa = \text{constant}$	κ adiabatic index/ exponent	1
Polytropic condition/ phase	$Pv^n = \text{constant}$	n polytropic index/ exponent	1
Equation of condition	$\frac{Pv}{T} = \text{constant}$	T absolute temperature	K
Ideal processes			
The Otto-process	$\eta_o = 1 - \frac{1}{\epsilon_n^{\kappa-1}}$	η_o ideal thermodynamic efficiency ϵ_n nominal degree of compression	1 1
Nominal degree of compression	$\epsilon_n = \frac{V_1}{V_2}$	V_1 cylinder-volume above piston in lower dead point V_2 cylinder-volume above piston in upper dead point	 m^3 m^3
Diesel process	$\eta_o = 1 - \frac{1}{\epsilon_n^{\kappa-1}} \cdot \frac{\rho^{\kappa-1}}{\kappa(\rho-1)}$	ρ volume condition during combustion	1
Adiabatic index	$\kappa = \frac{c_p}{c_v}$	c_p specific heat capacity at constant pressure c_v specific heat capacity at constant volume	 $\text{J/kg} \cdot \text{K}$ $\text{J/kg} \cdot \text{K}$

Volume condition during combustion

$$\Delta V = \frac{V_3}{V_2}$$

ΔV volume condition during combustion 1
 V_3 cylinder-volume above piston at the end of combustion m^3

Real processes

Pressure of compression

$$p_k = p_{s1} \epsilon^n$$

p_k pressure of compression Pa
 p_{s1} pressure at start of compression Pa
 ϵ degree of compression 1
 n politrop index 1

Compression temperature

$$T_k = T_{s1} \epsilon^{n-1}$$

T_k compression temperature K
 T_{s1} temperature at beginning of compression K

Efficiency and fuel consumption

consumption

Ideal thermodynamic efficiency

$$\eta_o = \frac{W_o}{Q_t}$$

η_o ideal thermodynamic efficiency 1
 W_o thermo work J
 Q_t supplied heat J

Indicated thermodynamic efficiency

$$\eta_i = \frac{W_j}{Q_t} = \frac{P_i}{\dot{Q}_t} = \frac{P_i}{\dot{m}_B h_g} = \frac{1}{b_i h_g}$$

η_i indicated thermodynamic efficiency 1
 W_1 work developed in cylinder (indicated work) J
 Q_1 supplied heat J
 P_i indicated effect W
 Q_t supplied energy per time unit J
 \dot{m}_B fuel consumption kg/s
 h_g upper fuel value J/kg
 b_i indicated specific fuel consumption kg/J

$$\eta_i = \eta_o \eta_g$$

η_g degree of goodness (inner efficiency) 1

Effective thermodynamic efficiency $\eta_e = \frac{W_e}{Q_t} = \frac{P_e}{\dot{Q}_t} = \frac{P_e}{\dot{m}_B h_g} = \frac{1}{b_e h_g}$

η_e effective thermodynamic efficiency 1
 W_e work supplied to motor shaft/axis (axis work) J
 P_e shaft/axis effect W
 b_e effective specific fuel consumption kg/J

$$\eta_e = \eta_i \eta_m$$

η_m mechanic efficiency 1

Degree of goodness

$$\eta_g = \frac{W_i}{W_o}$$

η_g degree of goodness 1
 W_o thermodynamic work J

Mechanical efficiency of engine

$$\eta_m = \frac{W_e}{W_i} = \frac{P_e}{P_i}$$

η_m mechanical efficiency of engine 1

Propeller shaft mechanical efficiency

$$\eta_a = \frac{W_p}{W_e} = \frac{P_p}{P_e}$$

η_a propeller shaft mechanical efficiency 1
 W_p propeller work J
 W_e shaft work J
 P_p propeller effect W

Fuel consumption

$$b_i = \frac{\dot{m}_B}{P_i}$$

b_i indicated specific fuel consumption kg/J

$$b_e = \frac{\dot{m}_B}{P_e}$$

b_e effective specific fuel consumption kg/J

Effect and mean pressure

Stroke volume per cylinder (displacement per cylinder)

$$V_h = \frac{\pi D^2}{4} S$$

V_h stroke volume per cylinder m^3
 D diameter of cylinder (bore) m
 S stroke m

Indicated effect of a two-stroke engine

$$P_i = i W_i n_a = i V_h \rho_{mi} n$$

P_i indicated effect W
 i number of cylinders 1
 W_i indicated work J
 n_a number of work processes per second and per cylinder s^{-1}
 ρ_{mi} indicated mean pressure Pa
 n frequency of rotation s^{-1}

$n_a = n$ for a two-stroke engine

Indicated effect of a four-stroke engine

$$P_i = i W_i n_a = i V_h \rho_{mi} \frac{n}{2}$$

$n_a = n/2$ for a four-stroke engine

Indicated mean pressure

$$\rho_{mi} = \frac{W_i}{V_h}$$

ρ_{mi} indicated mean pressure Pa

$$\rho_{mi} = \frac{\text{area of } pV\text{-diagram in mm}^2}{\text{length of } pV\text{-diagram in mm}} \quad (\text{scale factor of pressure shaft})$$

Shaft effect of two-stroke engine

$$P_e = i W_e n_a = i V_h \rho_{me} n$$

P_e shaft effect W
 W_e shaft work J
 ρ_{me} effective mean pressure Pa

$n_a = n$ for two-stroke engine

$$P_e = P_i \eta_m$$

η_m mechanic efficiency of engine 1

Shaft effect of four-stroke engine

$$P_e = i W_e n_a = i V_h \rho_{me} \frac{n}{2}$$

$$P_e = P_i \eta_m$$

$n_a = n/2$ for four-stroke engine

Equation for effective mean pressure

$$P_{me} = \frac{\rho_i h_g}{(L/B)_r} \frac{\eta_m \eta_i \gamma_f}{\lambda_f}$$

P_{me}	effective mean pressure	Pa
ρ	density of dry air outside engine	kg/m ³
h_g	oil's heating value	J/kg
$(L/B)_r$	reaction equivalent mixture of air and fuel	kg/kg
η_i	indicated thermo efficiency	1
γ_f	degree of air-filling of cylinder	1
λ_f	air factor related to air filling of cylinder	1

$$P_{me} = \eta_m P_{mi}$$

P_{mi}	indicated mean pressure	Pa
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$$T = \frac{P_e}{\omega} = \frac{P_e}{2\pi n}$$

Torque moment

T	torque moment	Nm
ω	angle velocity	rad/s

Mean piston velocity

$$c_{ms} = 2 S n$$

c_{ms}	mean piston velocity	m/s
S	length of stroke	m

COMBUSTION

Theoretic air demand

$$v_{Lr} = \frac{22.4}{0.21} \left(\frac{c}{12} + \frac{h}{4} + \frac{s}{32} \right)$$

v_{Lr}	theoretic need for air	$\frac{\text{Nm}^3 \text{ air}}{\text{kg fuel}}$
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Nm³ – normal cubic meter – not SI-unit

c	mass of carbon	kg
h	mass of hydrogen	kg
s	mass of sulfur	kg

$$(L/B)_r = 1.293 \cdot \frac{22.4}{0.21} \left(\frac{c}{12} + \frac{h}{4} + \frac{s}{32} \right)$$

$(L/B)_r$	theoretic need of air	$\frac{\text{kg air}}{\text{kg fuel}}$
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Real air usage

$$L/B = \lambda (L/B)_r$$

L/B	real air usage	$\frac{\text{kg air}}{\text{kg fuel}}$
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λ	air factor related to air usage	1
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Specific air usage	$l_e = b_e \lambda (L/B)_r$	l_e specific air usage	kg air/J
		b_e specific fuel usage	kg/J
Engine's air usage	$\dot{m}_L = P_e l_e = P_e b_e \lambda (L/B)_r$	\dot{m}_L air usage of engine	kg/s
		P_e shaft effect	W
	$\dot{m}_L = \dot{m}_B \lambda (L/B)_r$	\dot{m}_B fuel usage	kg/s
	$\dot{m}_L = \gamma_L i V_h \rho_i n_a$	γ_L degree of air usage	1
		i number of cylinders	1
		V_h displacement per cylinder	m ³
		ρ air density outside	kg/m ³
		n_a number of work processes per second	s ⁻¹

Exhaust gases	$V_{CO_2} = \frac{22.4 (c/12) \cdot 100}{22.4 (c/12) + v_{Lr} (\lambda - 0.21)}$	V_{CO_2} volume % CO ₂ (efficiency CO ₂) in dry exhaust	%
		c mass of carbon	kg

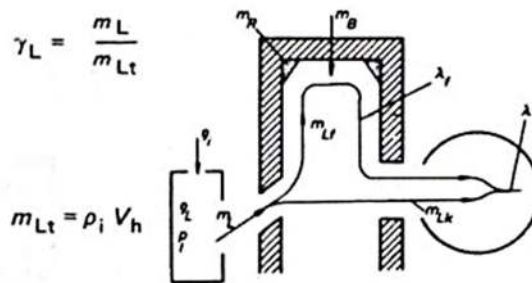
	$V_{O_2} = \frac{0.21 v_{Lr} (\lambda - 1) \cdot 100}{22.4 (c/12) + v_{Lr} (\lambda - 0.21)}$	V_{O_2} volume % O ₂ (efficiency O ₂) in dry exhaust	%
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	$V_{N_2} = \frac{0.79 v_{Lr} \lambda \cdot 100}{22.4 (c/12) + v_{Lr} (\lambda - 0.21)}$	V_{N_2} volume % N ₂ (efficiency N ₂) in dry exhaust	%
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Air factor	$\lambda = \frac{22.4 (c/12) (100 - V_{CO_2})}{V_{CO_2} v_{Lr}} + 0.21$	λ air factor related to air usage	1
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GAS CHANGING IN TWO-STROKE ENGINES

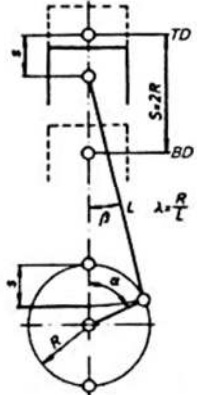
Degree of air usage	$\gamma_L = \frac{m_L}{m_{Lt}}$	γ_L degree of air usage	1
		m_L quantity of air used	kg
		m_{Lt} theoretic quantity of air filling	kg
		ρ density of suctioned air	kg/m ³
		V_h displacement (stroke volume)	m ³



Degree of air filling	$\gamma_f = \frac{m_{Lf}}{m_{Lt}} = \gamma_L (1 - \gamma_k)$	γ_f degree of air filling	1
		m_{Lf} quantity of air filled/ suctioned	kg
		γ_k degree of short circuit	1
Degree of scavenging air	$\gamma_R = \frac{m_{Lf}}{(m_{Lf} + m_R)}$	γ_R degree of scavenging air	1
		m_R quantity of residual gasses	kg
Air factor related to air filling	$\lambda_f = \lambda (1 - \gamma_k)$	λ_f air factor related to air filling	1
		λ air factor related to air usage	1
Degree of short circuit	$\gamma_k = \frac{m_{Lk}}{m_L}$	γ_k degree of short circuit	1
		m_{Lk} loss owing to short circuit + expelling	kg
	$m_L = m_{Lf} + m_{Lk}$	m_L quantity of air usage	kg

ENGINE DYNAMICS

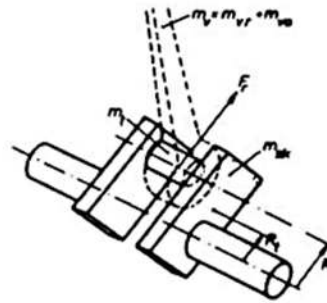
Crank-operation

Connecting rod ratio	$\lambda = \frac{R}{L}$		λ connecting rod ratio	1
			R crank radius	m
		L length of connecting rod	m	
Length of stroke	$s = R (1 - \cos \alpha + \frac{\lambda}{2} \sin^2 \alpha)$	s length of stroke	m	
		α angle of crank	rad	
Piston velocity	$c = R \omega (\sin \alpha + \frac{\lambda}{2} \sin 2\alpha)$	c piston velocity	m/s	
		ω angle velocity of crank	rad/s	
Piston acceleration	$a = R\omega^2 (\cos \alpha + \lambda \cos 2\alpha)$	a piston acceleration	m/s ²	
Piston acceleration in top and bottom dead centers	$a_{TD} = R\omega^2 (1 + \lambda)$	a_{TD} piston acceleration in top dead center	m/s ²	
	$a_{BD} = -R\omega^2 (1 - \lambda)$	a_{BD} piston acceleration in bottom dead center	m/s ²	

Oscillating force	$F_o = (m_s + m_{vo}) a$		F_o oscillating force	N
Gas force	$F_g = \frac{\pi d^2}{4} p$		F_s mass of whole piston	kg
Piston force	$F_p = F_g + F_o$	m_{vo} oscillating part of mass of connecting rod	kg	
Normal force	$F_n = F_s \operatorname{tg} \beta$	F_g gas force	N	
Connecting rod force	$F_v = \frac{F_p}{\cos \beta}$	d piston diameter	m	
Tangential force	$F_t = F_v \sin (\alpha + \beta)$	p difference of pressure on upper and lower side of piston	Pa	
Work per revolution	$F_t = \frac{F_p \sin (\alpha + \beta)}{\cos \beta}$	F_p piston force	N	
Mean indicated torque	$W_i = F_{tm} 2 \pi R$	F_n normal force	N	
Radial force	$T_i = F_{tm} R$	β angle of connecting rod	rad	
Rotating force	$F_d = F_v \cos (\alpha + \beta)$	F_v connecting rod force	N	
	$F_r = m_r R \omega^2$	F_t tangential force	N	
		W_i work	J	
		F_{tm} mean tangential force	N	
		T_i torque	N · m	
		F_d radial force	N	
		F_r rotating force	N	
		m_r mass of crank journals/pin + mass of counterweights related to crank radius + rotating part of connecting rod mass m_{vr}	kg	

Mass forces in a one-cylinder engine

Rotating mass forces $F_r = m_{vr} R \omega^2 + m_t R \omega^2 + 2 m_{sk} R_t \omega^2$



$$F_r = m_r R \omega^2$$

$$m_r = m_{vr} + m_t + 2m_{sk} \frac{R_t}{R}$$

F_r	rotating mass forces	N
m_{vr}	rotating mass part of connecting rod	kg
R	crank radius	m
ω	angle of velocity	rad/s
m_t	mass of the crankshaft pin	kg
m_{sk}	mass of the crank counterweight	kg
R_t	the radius of the center of gravity of the counterweight	m

m_r	mass of crankshaft/ journal pin + mass of counterweights related to crank radius + rotating part of connecting rod mass m_{vr}	kg
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Oscillating mass forces $F_o = m_o R \omega^2 (\cos \alpha + \lambda \cos 2\alpha)$

F_o	oscillating mass forces	N
m_o	total oscillating mass	kg
α	angle of crank	rad
λ	connecting rod ratio	1

$$m_o = m_p + m_{vo}$$

m_p	mass of complete piston	kg
m_{vo}	oscillating mass of connecting rod	kg

$$F_{01} = m_o R \omega^2 \cos \alpha$$

F_{01}	mass forces of first order	N
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$$F_{02} = \lambda m_o R \omega^2 \cos 2\alpha$$

F_{02}	mass forces of second order	N
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Centrifugal force of counter weights

$$F_k = F_r + \frac{1}{2} F_{01 \text{ maximum}}$$

F_k	counter weights' centrifugal force	
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Two counter weights

$$2m_k R_k \omega^2 = m_r R \omega^2 + \frac{1}{2} m_o R \omega^2$$

m_k	mass per counter weight	kg
R_k	point of gravity radius	m
m_r	rotating mass	kg
m_o	oscillating mass	kg

Torque swings

Swing frequency $f_{n1} = \frac{1}{2\pi} \sqrt{\frac{G I_p}{L} \cdot \frac{I_1 + I_2}{I_1 \cdot I_2}}$

f_{n1}	frequency	Hz
G	shearing module of shaft	N/m ²
I_p	polar inertia moment of shaft	m ⁴
L	shaft length	m
I_1	inertia moment of flywheels	kg/m ²
I_2	inertia moment of flywheels	kg/m ²

Plane pressure in bearings

$$p = \frac{\frac{\pi}{4} d_v^2 \cdot p_f}{D_v L_{ve}}$$

p	bearing pressure	Pa
d_v	cylinder diameter	m
p_f	maximum combustion pressure	Pa
D_v	diameter of bearing journal	m
L_{ve}	effective bearing length (with deduction for eventual grease tracks)	m

PROPULSION IN WATER

Effect needed

Propulsion effect (towing effect) $P_E = F_T v$

P_E	propulsion effect	W
F_T	propulsion resistance	N
v	ship velocity	m/s

Pushing power/force $F_p = F_T + F_M$

F_p	pushing force	N
F_M	additional resistance from propeller	N

Propeller's delivered effect (pushing effect) $P_s = F_s v_r$

P_s	pushing effect	W
v_r	propeller's velocity in propulsion direction related to water	m/s

$$\text{current coefficient} = \frac{v - v_r}{v}$$

v_r	relative velocity of propeller	
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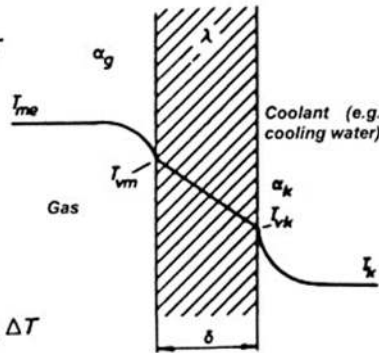
Current coefficient 0.20 – 0.25

Propeller efficiency	$\eta_p = \frac{P_s}{P_p}$	η_p propeller efficiency P_p effect supplied to propeller	1 W
Propulsion efficiency	$\eta_f = \frac{P_E}{P_p} = \frac{P_E}{P_s} \frac{P_s}{P_p} = \eta_{sg} \eta_p$	η_f propulsion efficiency	1
	$\eta_{sg} = \frac{P_E}{P_s}$	η_{sg} shape of hull (degree of goodness of hull)	1
Shaft mechanical efficiency	$\eta_a = \frac{P_p}{P_e}$	η_a shaft mechanical efficiency P_e shaft effect from engine	1 W
Ship's theoretical speed	$v_t = s n_p$	v_t ship's theoretical velocity s propeller thread/rise n_p propeller rotating frequency	m/s m s ⁻¹
Slip	$slip = \frac{(v_t - v)}{v_t} 100$		
	$P_e = P_e^* \left(\frac{n}{n^*}\right)^3 = P_e^* \left(\frac{v}{v^*}\right)^3$	P_e shaft effect P_e^* shaft effect at normal operation n rotation frequency n^* rotation frequency at normal operation v^* velocity at normal operation	W W s ⁻¹ s ⁻¹ m/s
	$P_e = \Delta^{\frac{2}{3}} \frac{v^3}{A}$	Δ displacement A admiralty constant	ton

HEAT TRANSFER AND HEAT LOAD

Heat flow

$$\dot{Q} = \alpha_g A \Delta T$$



Heat flux

$$\dot{q} = \frac{\dot{Q}}{A} = \alpha_g \Delta T$$

Eichelberg's formula

$$\alpha_g = 2.1 \sqrt{\rho T} \sqrt[3]{c_{ms}}$$

One-dimensional transfer

$$\dot{q}_m = \frac{\lambda}{\delta} (T_{vm} - T_{vk})$$

Heat transfer through wall

$$\dot{q}_m = k (T_{me} - T_k)$$

\dot{Q}	heat flow	J/s = W
α_g	heat transfer coefficient	W/(m ² · K)
A	area to which heat is transferred	m ²
ΔT	temperature difference between gas and surface material	K

\dot{q} heat flux (density of heat flow)

α_g	heat transfer coefficient	kcal/m ² h
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Eichelberg's formula is based on the technical system.

ρ	absolute gas pressure	kp/cm ²
T	gas temperature	K
c_{ms}	mean piston velocity	m/s

\dot{q}_m	mean heat flux	W/m ²
λ	thermo conductivity	W/(m ² · K)
δ	material thickness	m
T_{vm}	mean wall temperature on gas side	K
T_{vk}	mean wall temperature on refrigerant side	K

α_k	heat transfer coefficient	W/(m ³ · K)
T_k	refrigerant temperature	K

k	heat transfer coefficient	W/(m ³ · K)
T_{me}	mean effective gas temperature	K

Heat transfer coefficient

$$\frac{1}{k} = \frac{1}{\alpha_{gm}} + \frac{\delta}{\lambda} + \frac{1}{\alpha_k}$$

α_k heat transfer coefficient gas-surface $W/(m^2 \cdot K)$

α_k heat transfer coefficient material-refrigerant $W/(m^2 \cdot K)$

Any boiler scale shall be added $\frac{\delta_{ks}}{\lambda_{ks}}$.

AUXILIARY SYSTEMS

Cooling water system

Necessary quantity of cooling water

$$\dot{v} = \frac{\sum r_k}{\Delta T_{tot} c \rho \eta_e}$$

\dot{v} specific cooling water quantity m^3/J

$\sum r_k$ heat quantity transferred in cooler, in percentage of supplied energy %

ΔT_{tot} temperature difference K

c specific heat capacity $J/(kg \cdot K)$

ρ density kg/m^3

η_e effective thermo efficiency 1

$$\dot{v} = \frac{\sum r_k \dot{m}_B h}{\Delta T_{tot} c \rho P_e}$$

\dot{m}_B fuel consumption kg/s
 h calorific value J/kg

$$\dot{V} = \dot{v} P_e$$

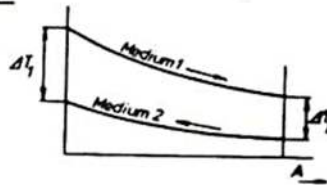
\dot{V} water quantity m^3/s
 P_e shaft efficiency W

Freshwater coolers

Heat flow	$\dot{Q} = r_{kf} \dot{m}_B h = r_{kf} h b_e P_e = r_{kf} \frac{P_e}{\eta_e}$	\dot{Q} heat flow r_{kf} heat quantity carried off in freshwater coolers in percentage of supplied energy \dot{m}_B fuel consumption h calorific value b_e effective specific fuel consumption P_e shaft efficiency η_e effective thermo efficiency	J/s = W % kg/s J/kg kg/J W 1
	$\dot{Q} = k A \Delta T$	k heat transfer coefficient A cooling area ΔT logarithmic mean temperature difference	W/(m ² · K) m ² K

Logarithmic mean temperature difference

$$\Delta T = \frac{\Delta T_1 - \Delta T_2}{\ln \frac{\Delta T_1}{\Delta T_2}}$$



ΔT logarithmic mean temperature difference	K
ΔT_1 temperature difference	K
ΔT_2 temperature difference	K

Lubricating coolers

Heat flow	$\dot{Q} = r_{ks} \dot{m}_B h = r_{ks} h b_e P_e = r_{ks} \frac{P_e}{\eta_e}$	\dot{Q} heat flow r_{ks} transferred heat quantity in lubricating coolers in percentage of supplied energy P_e shaft effect	J/s = W % W
Effect needed for lubricating pump	$P = \frac{\Delta p \dot{V}}{\eta} = \frac{\Delta p \dot{v} P_e}{\eta}$	P effect needed Δp increased pressure in pump \dot{V} oil quantity η pump efficiency \dot{v} oil quantity	W Pa m ³ /s 1 m ³ /J